

COURSE: Aerodynamics		
DEGREE: Aerospace Engineering	YEAR: 3rd	TERM: 1st

WEEKLY PLANNING							
WEEK	SESSION	DESCRIPTION	TEACHING (mark X)		SPECIAL ROOM FOR SESSION (Computer class room, audio-visual class room)	WEEKLY PROGRAMMING FOR STUDENT	
			LECTURES	SEMINARS		DESCRIPTION	CLASS HOURS (1,66=50+50 min) HOMEWORK HOURS (Max. Estim. 6,5h)
1	1	Review of Fluid Mechanics: - Euler equations - Subsonic and supersonic flows - Irrotational flows	x				1,66
	2	Aerodynamic Forces and Moments Problem Sheet #1		x			1,66
2	3	Generation of lift in 2D (1/2) - Complex potential - Elementary solutions - Non-lifting flow around a cylinder	x				1,66
	4	Problem Sheet #2		x			1,66
3	5	Generation of lift in 2D (2/2) - Flow around a cylinder with circulation - Kutta-Joukowski theorem - D'Alemberts Paradox - Kutta's condition and the starting vortex	x				1,66
	6	Problem Sheet #3		x			1,66
4	7	Incompressible flow over airfoils (1/4) - Sheet of vortices and sheet of sources - Linearization of the boundary conditions QUIZ #1 (50 min)	x				1,66
	8	LABORATORY #1: Panel method MATLAB		x	online		1,66
5	9	Incompressible flow over airfoils (2/4) - Asymmetric problem: angle of attack and camber	x				1,66
	10	Problem Sheet #4		x			1,66
6	11	Incompressible flow over airfoils (3/4) - Symmetric problem: thickness - Drag and Stall	x				1,66
	12	Problem Sheet #5		x			1,66
7	13	Incompressible flow over airfoils (4/4) - High lift devices Incompressible flow over finite wings (1/4) - Biot Savart law	x				1,66
	14	LABORATORY #2: Panel method XFRLS		x	online		1,66
8	15	Incompressible flow over finite wings (2/4) - Prandtl's lifting line theory QUIZ #2 (50 min)	x				1,66
	16	Problem Sheet #6		x			1,66
9	17	Incompressible flow over finite wings (3/4) - Elliptic lift distribution - General lift distribution	x				1,66
	18	Problem Sheet #7		x			1,66
10	19	Incompressible flow over finite wings (4/4) - Initial and unitary lift distribution	x				1,66
	20	Problem Sheet #8		x			1,66
11	21	Compressibility effects in 2D airfoils - Linearization of the equations QUIZ #3 (50 min)	x				1,66
	22	LABORATORY #3: Re-design of the CN212-400 wing		x	online		1,66
12	23	Linearized theory for subsonic airfoils - Prandtl-Glauert compressibility correction - Critical Mach number - Supercritical airfoil	x				1,66
	24	Problem Sheet #9		x			1,66
13	25	Linearized theory for supersonic airfoils (1/2) - Drag Divergence - Linearized theory for supersonic airfoils	x				1,66
	26	Problem Sheet #10		x			1,66
14	27	Linearized theory for supersonic Airfoils (2/2) - Aerodynamic interference in supersonic airfoils - Area Rule	x				1,66
	28	Problem Sheet #11		x			1,66
29		LABORATORY SESSION #4: Wind tunnel		x	online		1,66
Subtotal 1						48	94

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Total 1 (Hours of class plus student homework)							142	
15		Tutorials, handing in, etc					3,6	-
16	Assessment						4	10
17								
18								
Subtotal 2							8	10
Total 2 (Hours of class plus student homework)							18	
TOTAL (Maximun 160 horas)							160	