

<b>COURSE: Aerodynamics</b>		
<b>DEGREE: Aerospace Engineering</b>	<b>YEAR: 3rd</b>	<b>TERM: 1st</b>

WEEKLY PLANNING								
WEEK	SESSION	DESCRIPTION	TEACHING (mark X)		SPECIAL ROOM FOR SESSION (Computer class room, audio-visual class room)	WEEKLY PROGRAMMING FOR STUDENT		
			L E C T U R E S	S E M I N A R S		DESCRIPTION	CLASS HOURS (1,66=50+50 min)	HOMEWORK HOURS (Max. Estim. 6,5h)
1	1	Review of Fluid Mechanics: - Euler equations - Subsonic and supersonic flows - Irrotational flows	X				1,66	6,5
	2	Aerodynamic Forces and Moments Problem Sheet #1		X			1,66	
2	3	Generation of lift in 2D (1/2) - Complex potential - Elementary solutions - Non-lifting flow around a cylinder	X				1,66	6,5
	4	Problem Sheet #2		X			1,66	
3	5	Generation of lift in 2D (2/2) - Flow around a cylinder with circulation - Kutta-Joukowski theorem - D'Alamberts Paradox - Kutta's condition and the starting vortex	X				1,66	6,5
	6	Problem Sheet #3		X			1,66	

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4	7	Incompressible flow over airfoils (1/4) - Sheet of vortices and sheet of sources - Linearization of the boundary conditions QUIZ #1 (50 min)	x			1,66	6,5	
	8	LABORATORY #1: Panel method MATLAB		x	computer	1,66		
5	9	Incompressible flow over airfoils (2/4) - Asymmetric problem: angle of attack and camber	x			1,66	6,5	
	10	Problem Sheet #4		x		1,66		
6	11	Incompressible flow over airfoils (3/4) - Symmetric problem: thickness - Drag and Stall	x			1,66	6,5	
	12	Problem Sheet #5		x		1,66		
7	13	Incompressible flow over airfoils (4/4) - High lift devices Incompressible flow over finite wings (1/4) - Biot Savart law	x			1,66	6,5	
	14	LABORATORY #2: Panel method XFLR5		x	computer	1,66		
8	15	Incompressible flow over finite wings (2/4) - Prandtl's lifting line theory QUIZ #2 (50 min)	x			1,66	6,5	
	16	Problem Sheet #6		x		1,66		
9	17	Incompressible flow over finite wings (3/4) -Elliptic lift distribution -General lift distribution	x			1,66	6,5	
	18	Problem Sheet #7		x		1,66		
10	19	Incompressible flow over finite wings (4/4) - Initial and unitary lift distribution	x			1,66	6,5	
	20	Problem Sheet #8		x		1,66		

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11	21	Compressibility effects in 2D airfoils - Linearization of the equations QUIZ #3 (50 min)	x				1,66	6,5
	22	LABORATORY #3: Re-design of the CN212-400 wing		x	computer		1,66	
12	23	Linearized theory for subsonic airfoils - Prandtl-Glauertc compressibility correction - Critical Mach number - Supercritical airfoil	x				1,66	6,5
	24	Problem Sheet #9		x			1,66	
13	25	Linearized theory for supersonic airfoils (1/2) - Drag Divergence - Linearized theory for supersonic airfoils	x				1,66	6,5
	26	Problem Sheet #10		x			1,66	
14	27	Linearized theory for supersonic Airfoils (2/2) - Aerodynamic interference in supersonic airfoils - Area Rule	x				1,66	6,5
	28	Problem Sheet #11		x			1,66	
	29	LABORATORY SESSION #4: Wind tunnel			7.0.H03		1,66	3,25
<b>Subtotal 1</b>							<b>48</b>	<b>94</b>
<b>Total 1 (Hours of class plus student homework)</b>							<b>142</b>	
15		Tutorials, handing in, etc					3,6	-
16		Assessment					4	10
17								
18								
<b>Subtotal 2</b>							<b>8</b>	<b>10</b>
<b>Total 2 (Hours of class plus student homework)</b>							<b>18</b>	

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			L E C T U R E S	S E M I N A R S		DESCRIPTION	CLASS HOURS (1,66=50+50 min)	HOMEWORK HOURS (Max. Estim. 6,5h)
<b>TOTAL</b> ( <i>Maximun 160 horas</i> )						<b>160</b>		