

Academic Year: ( 2017 / 2018 )

Review date: 05-09-2017

Department assigned to the subject: Department of Bioengineering and Aerospace Engineering

Coordinating teacher: HILARIO MONTES, JAVIER

Type: Compulsory ECTS Credits : 6.0

Year : 2 Semester : 1

**STUDENTS ARE EXPECTED TO HAVE COMPLETED**

Advanced Aircraft Design and Certification II  
 Advanced Aeroelasticity  
 Propulsion Systems Performance and Design

It is needed to have a good knowledge, at grade-level, of:  
 Aerodynamics  
 Flight Mechanics

**COMPETENCES AND SKILLS THAT WILL BE ACQUIRED AND LEARNING RESULTS.****COMPETENCES**

Capacity to project, build, inspect, certify and maintain any type of aircraft, including its particular subsystems.  
 Aptitude for projecting, building, inspecting, certifying and maintaining any type of aircraft. Necessary knowledge and capacities to perform the analysis and structural design of aircrafts, including skills in advanced aircraft structural design and analysis SW tools.  
 Capacity to design, execute and analyze Flight and Ground Tests of Aircrafts.  
 Capacity to carry out the complete process of Aircraft Certification.  
 Suitable knowledge of the different Aircraft Subsystems.

**EDUCATIONAL ACHIEVEMENTS**

When concluding the course the student must be capable of:  
 Sizing the aircraft based on its mission and the adequate flying qualities criteria.  
 Calculating the design structural loads acting on the aircraft, as well as the fatigue conditions. Based on this, to estimate the structural and operational weights.  
 Knowing the aircraft certification requirements and specifying the means of compliance associated to them, judging their levels of acceptance.  
 Specifying the means or tests necessary to demonstrate that the aircraft fulfills the certification requirements.  
 Knowing and specifying and/or designing the subsystems of the aircrafts.  
 Evaluating the operational costs of the aircraft.  
 Evaluating the interaction between systems, flying qualities and structural loads, including failure cases.  
 Knowing the specific design constraints of special aircraft types.

**DESCRIPTION OF CONTENTS: PROGRAMME**

Aircraft Sizing Review --> Aircraft Geometry, Quick Mass Sizing and P/L-Range, T/W and Wing Loading.

Longitudinal Flying Qualities and HTP design --> Introduction & common A/C tail configurations, HTP main functions, Tail architecture, Basic design guidelines, Longitudinal stability and control, Horizontal tail sizing in preliminary design.

Latero-Directional Flying Qualities and VTP design --> VTP main functions & common A/C tail configurations, VTP architecture, Basic design guidelines, Lateral-directional stability and control, Tail sizing in preliminary design.

Powerplant selection and installation --> Engine Types, Engine Integration, Propeller Analysis, Engine Ratings.

Landing Gear design --> Wheel suspension system, Landing gear retraction, Geometric design criteria, Equivalent single wheel load, Runway classification: LCN/LCG system.

Flight Loads --> General Introduction to Loads, Design Envelopes, Flight Load Conditions, NZ manoeuvres and Discrete Gusts, Pitch manoeuvres, Roll manoeuvres, Yaw and OEO manoeuvres.

Ground Loads --> Ground Loads Conditions, Landing Loads, Ground Handling Loads, General Aspects of Ground Loads.

Fatigue analysis --> Fatigue concept, Infinite-life, Safe-life, Fail-safe, Damage Tolerant: Accidents that changed aerospace engineering, The fatigue design Philosophy, Fatigue Analysis.

Weight estimation --> Aircraft Mass Breakdown, Rapid Mass Estimation Method, Semi-empirical Mass Estimation Method, CG and Moments of Inertia

Interaction Systems-Flight-Structures --> Aircraft Systems, Analysis of Systems Failures

Aircraft Cost analysis --> Life cycle cost analysis, Indirect operating cost, Direct operating cost.

Flight and Ground Testing --> Scope, Ground Tests, Flight Tests.

#### LEARNING ACTIVITIES AND METHODOLOGY

Theory sessions.

Problem sessions working individually and in groups.

Lab sessions in computer rooms.

#### ASSESSMENT SYSTEM

Continuous evaluation (40%)

End-of-term exam (60%)

In order to pass the subject a mark of 5.0/10 (Continuous evaluation + End-of-term exam) must be obtained. It is necessary to have a minimum mark of 4.0/10 in the end-of-term exam to pass the subject.

**% end-of-term-examination:** 60

**% of continuous assessment (assignments, laboratory, practicals...):** 40

#### BASIC BIBLIOGRAPHY

- Ajoy Kumar Kundu Aircraft Design, Cambridge Aerospace Series, 2010
- Daniel P. Raymer Aircraft Design: A Conceptual Approach, AIAA Education Series, 2012
- Jan Roskam Airplane Design I-VIII, DAR corporation, 1985-1990
- Ralph D. Kimberlin Flight Testing of Fixed-Wing-Aircraft, AIAA Education Series, 2003
- Ted L. Lomax Structural Loads Analysis for Commercial Transport Aircraft , AIAA Education Series, 1996

#### ADDITIONAL BIBLIOGRAPHY

- Denis Howe Aircraft Conceptual Design Synthesis, Professional Engineering Publishing Limited, 2000
- Denis Howe Aircraft Loading and Structural Layout, AIAA Education Series, 2004
- Egbert Torenbeek Synthesis of Subsonic Airplane Design, Delft University Press, 1982
- Leland M. Nicolai, Grant E. Carichner Fundamentals of Aircraft and Airship Design, Vol 1 - Aircraft Design, AIAA Education Series, 2010
- Lloyd R. Jenkinson, Paul Simpkin, Darren Rhodes Civil Jet Aircraft Design, AIAA Education Series, 1999
- Steven A. Brandt; Randall J. Stiles; John J. Bertin; Ray Whitford Introduction to Aeronautics: A Design Perspective, AIAA Education Series, 2015